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# Human Factors Implications of Continuous Descent Approach Procedures for Noise Abatement in Air Traffic Control

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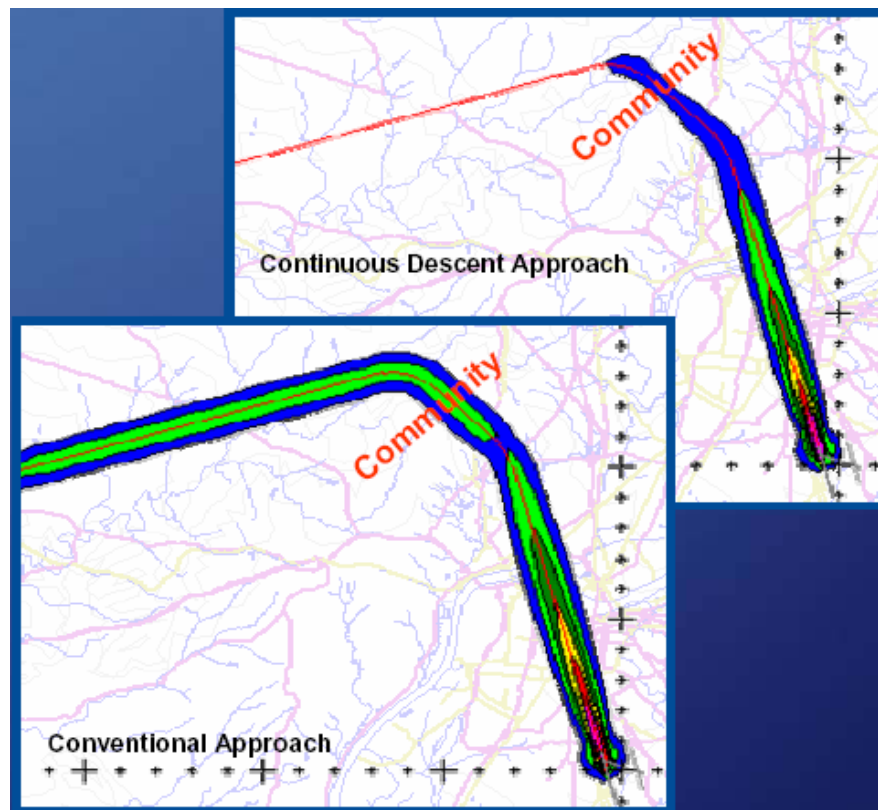
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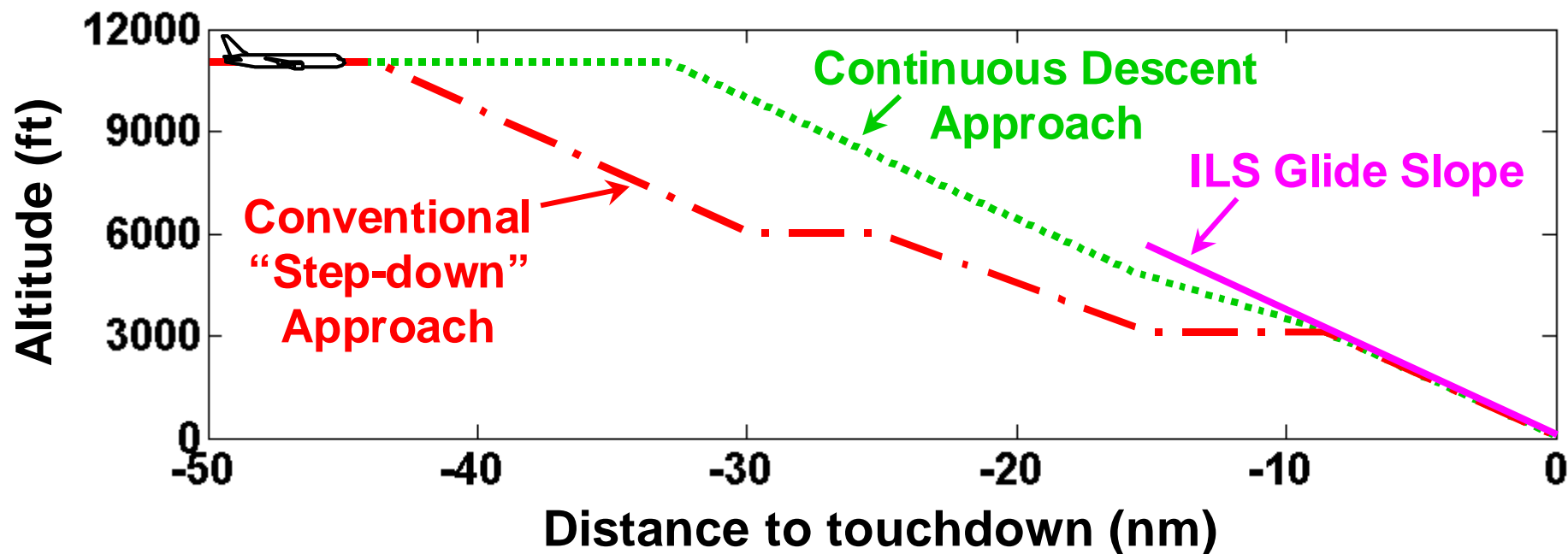
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**MIT International Center for Air Transportation &  
Cambridge/MIT Institute Silent Aircraft Initiative**

# Why use CDA Procedures?

- ❑ Noise impact on communities is one of the major limitations to air transportation infrastructure expansion
- ❑ Noise target levels require improvements in both **aircraft design + operational procedures**
- ❑ Continuous Descent Approach (CDA) procedures can reduce noise exposure by 3-6.5 dBA (3dB= 50% acoustic energy reduction)



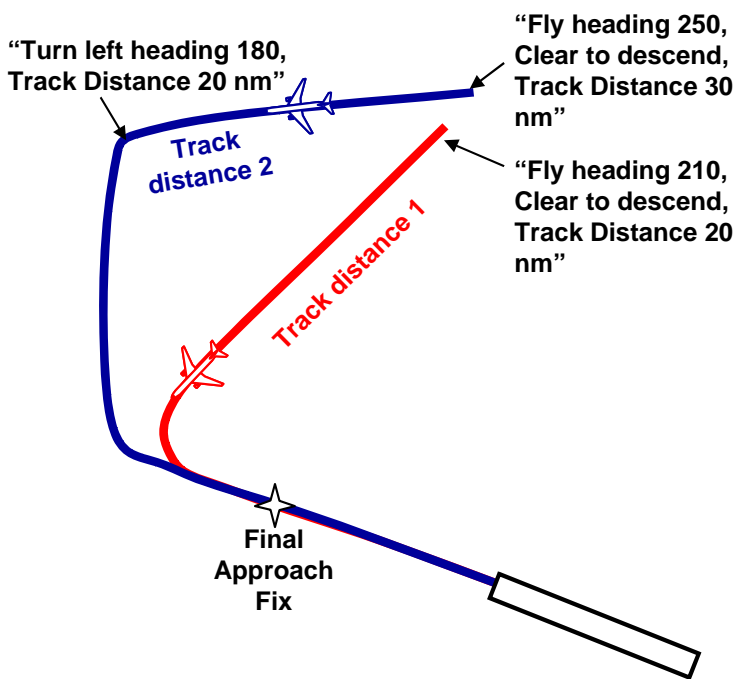


- Keep aircraft higher and at lower thrust for longer than conventional approach

# CDA Concept- Vertical/Lateral coupling

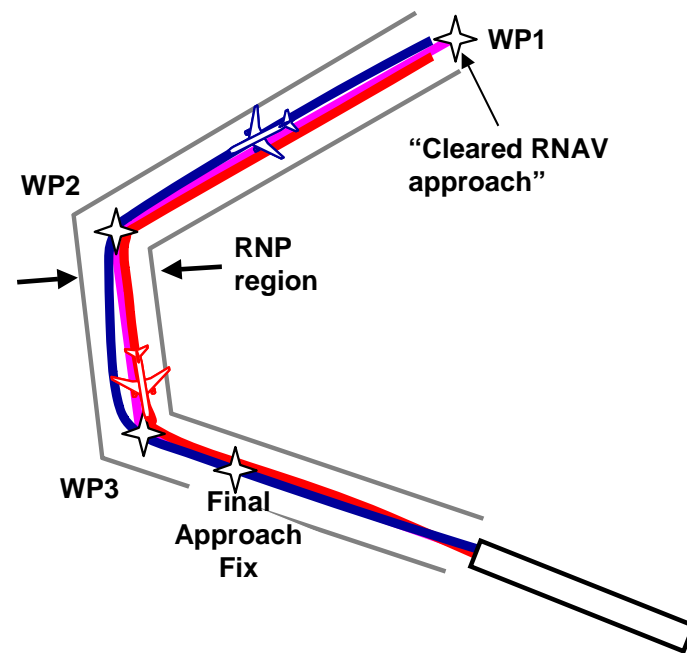
## □ Basic CDA

- Controller:
  - Retains lateral & speed control
  - Provides track distance estimate
- Pilot:
  - Estimates descent rate using track distance



## □ RNAV CDA

- Controller:
  - Clears aircraft for RNAV approach
- Pilot & FMS:
  - Programmed approach
  - Optimized descent rate using altitude targets and speed





# Motivation for Investigation of CDA Human Factors Issues

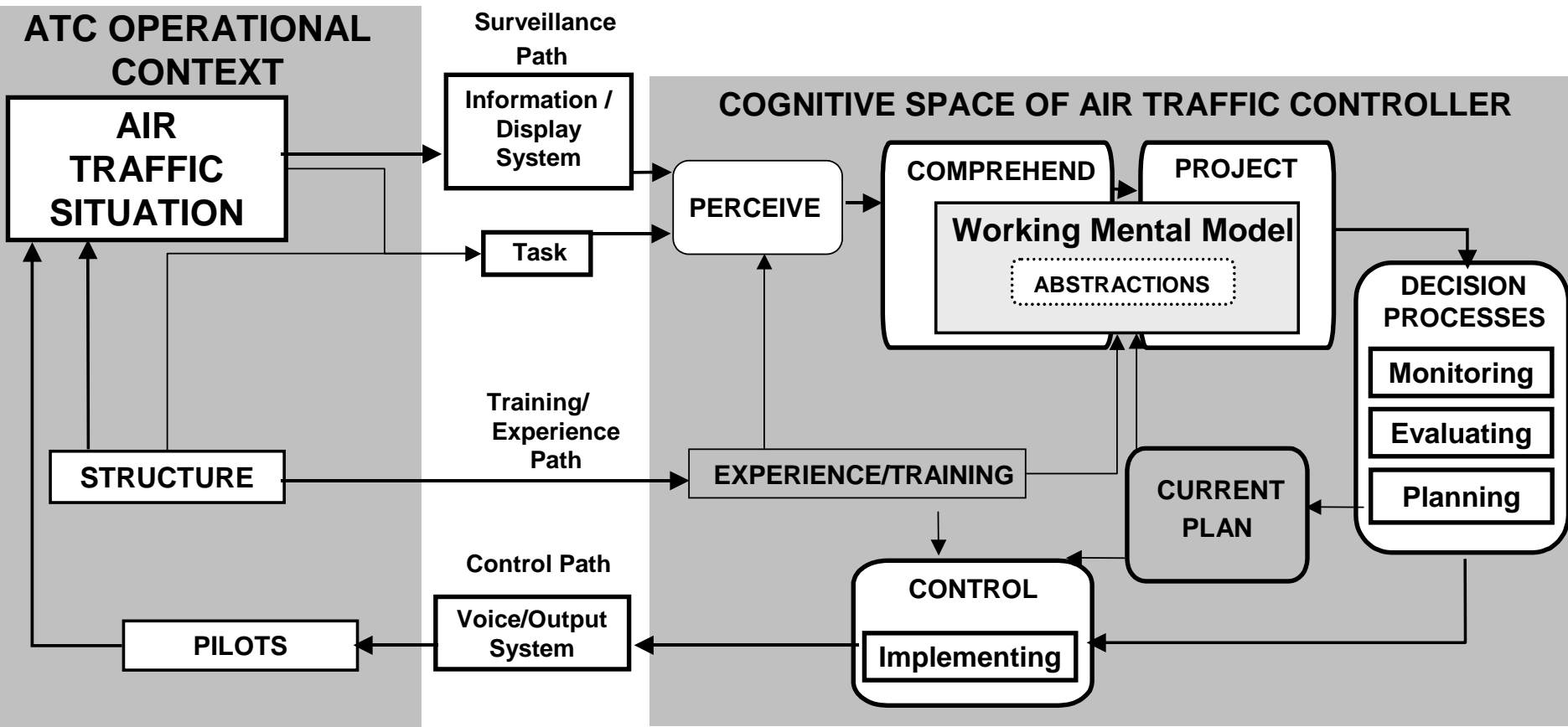
- CDA procedures are significantly different than the conventional approach currently in use and have implications on controller cognitive processes, including projection
- Implementation of the CDA procedures may present challenges with approach operations:
  - Traffic throughput
    - 50% throughput reduction compared with conventional approach in trials conducted by Clarke, Ho, & Ren, 2004
  - Controller acceptance of procedure (effect on cognitive processes)
  - Controller workload



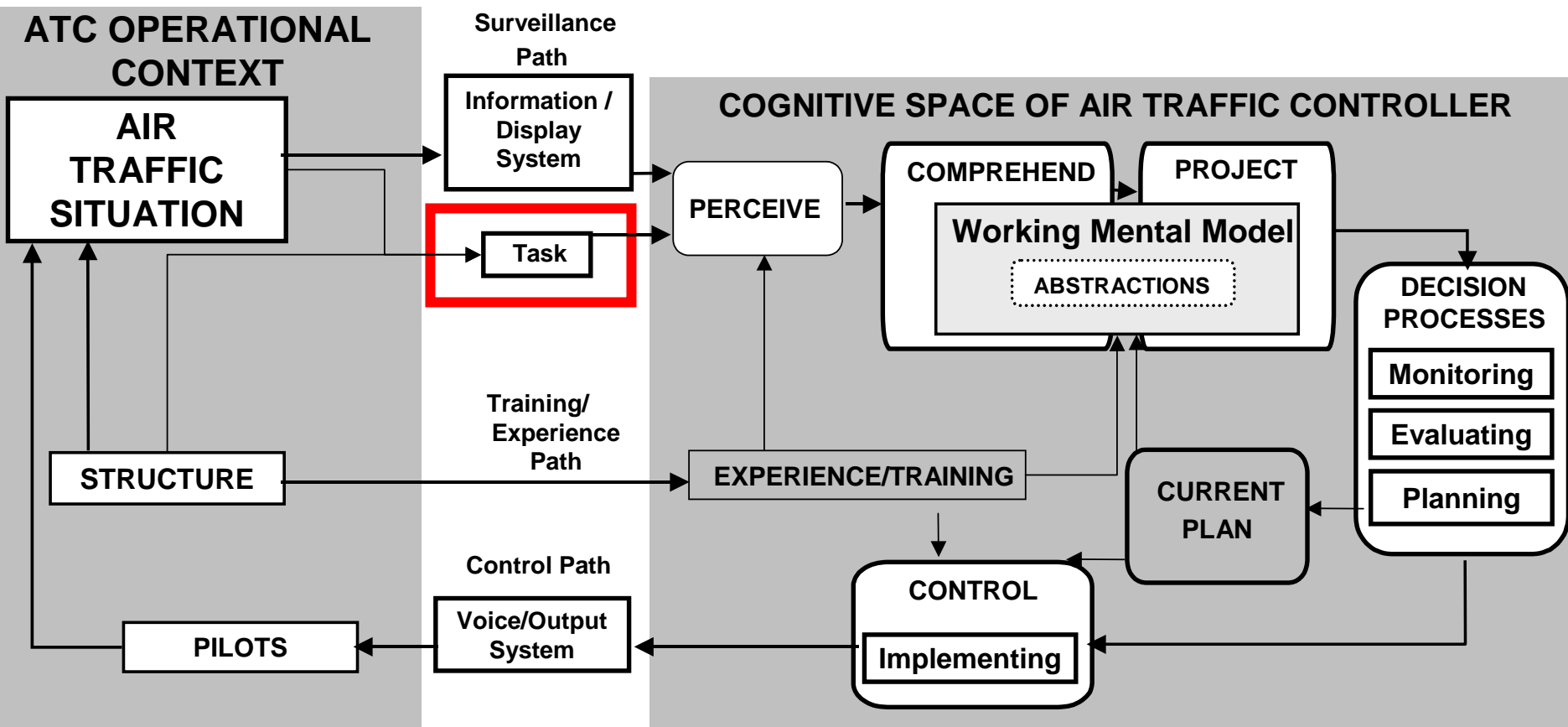
# Approach

- ❑ ATC Process Model Development
  - Incorporates Endsley's Situation Awareness Model & Pawlak's Decision Processes model
  - Modified based on U.S. ATC 7110.65 & Boston & NY SOPs
  - Site visits used to revise model (Boston, NY, Manchester, Reykjavik)
- ❑ Application of ATC Process Model to Final Approach Task
- ❑ Cognitive Difference Analysis performed using ATC Process Model as a means to identify cognitive issues with CDA procedures
- ❑ Experiment performed testing utility of an identified key approach abstraction
- ❑ CDA procedure implementation guidance provided based on results

# ATC Process Model

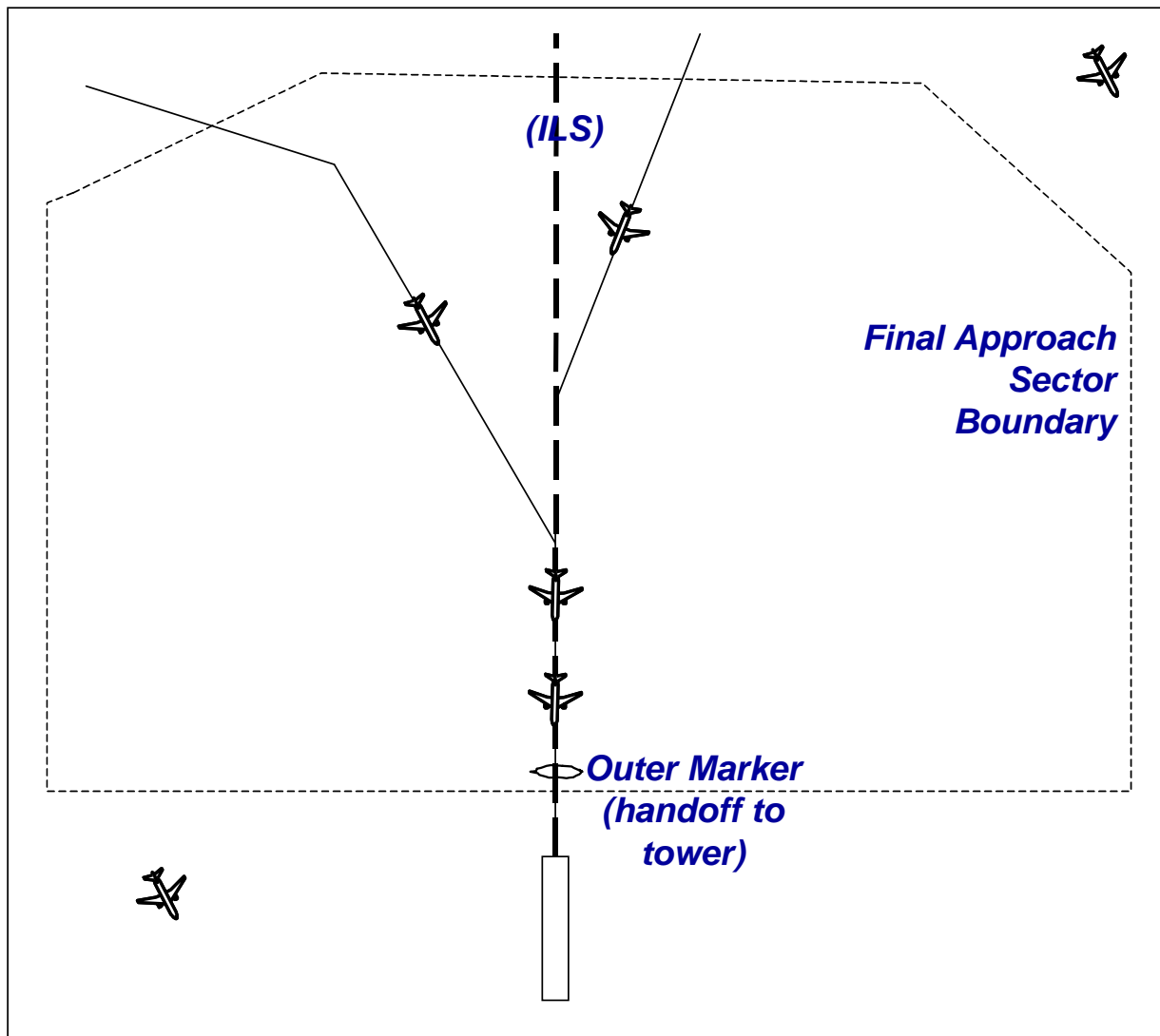


# Controller's Task





# Final Approach Control Task



**Vector aircraft onto approach (laterally & vertically)**

**Manage separation:**

- **Compress traffic in periods of high demand**

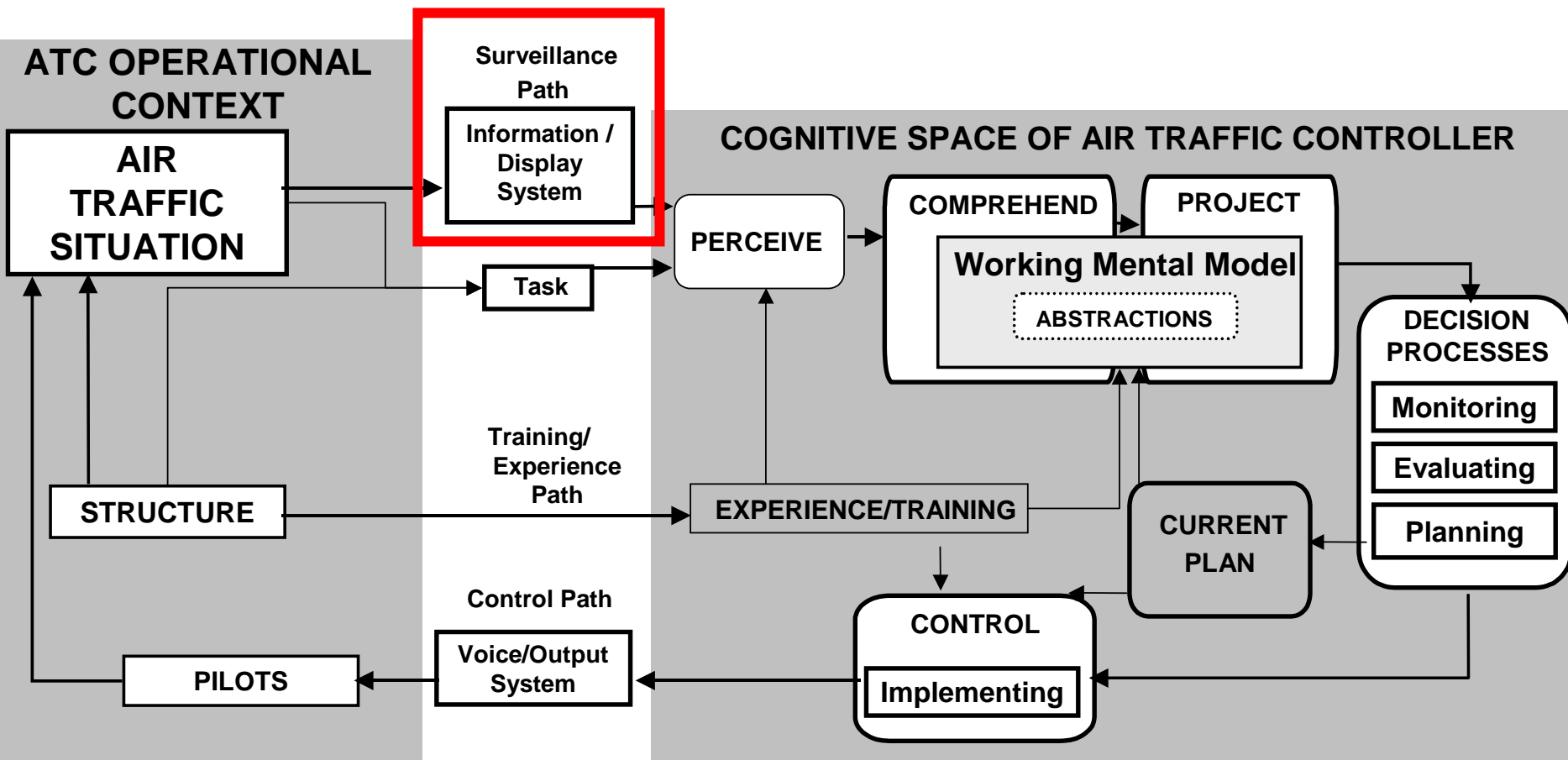
- **Ensure minimum separation**

- **1000 ft vertical, OR**

- **3-6 nm (wake vortex) longitudinal**

**Other tasks**

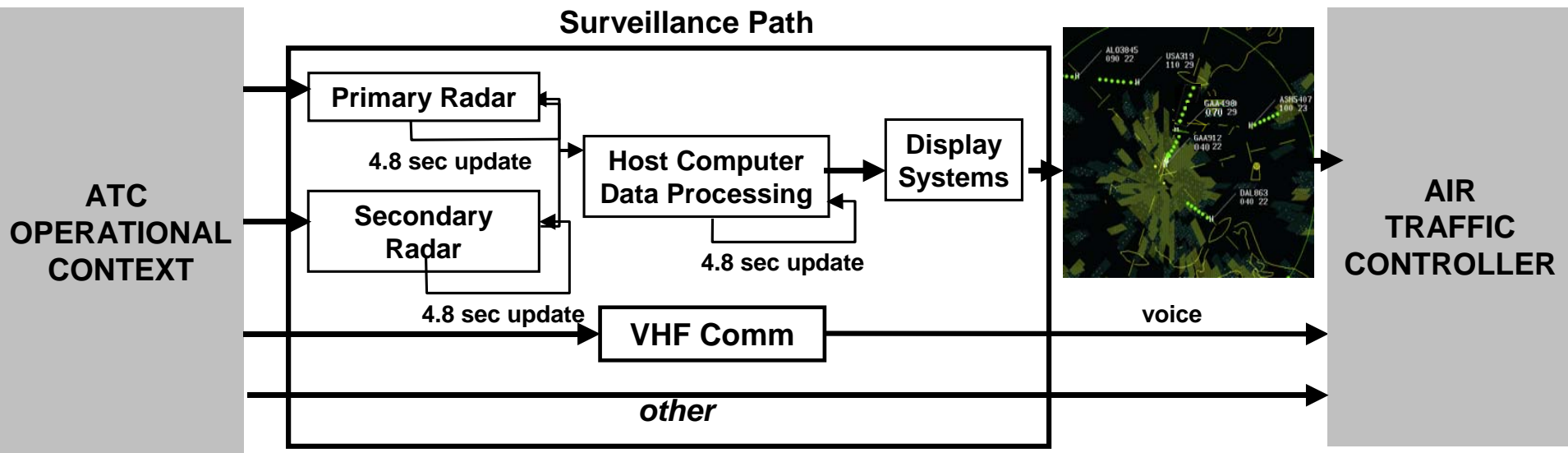
# Surveillance Path



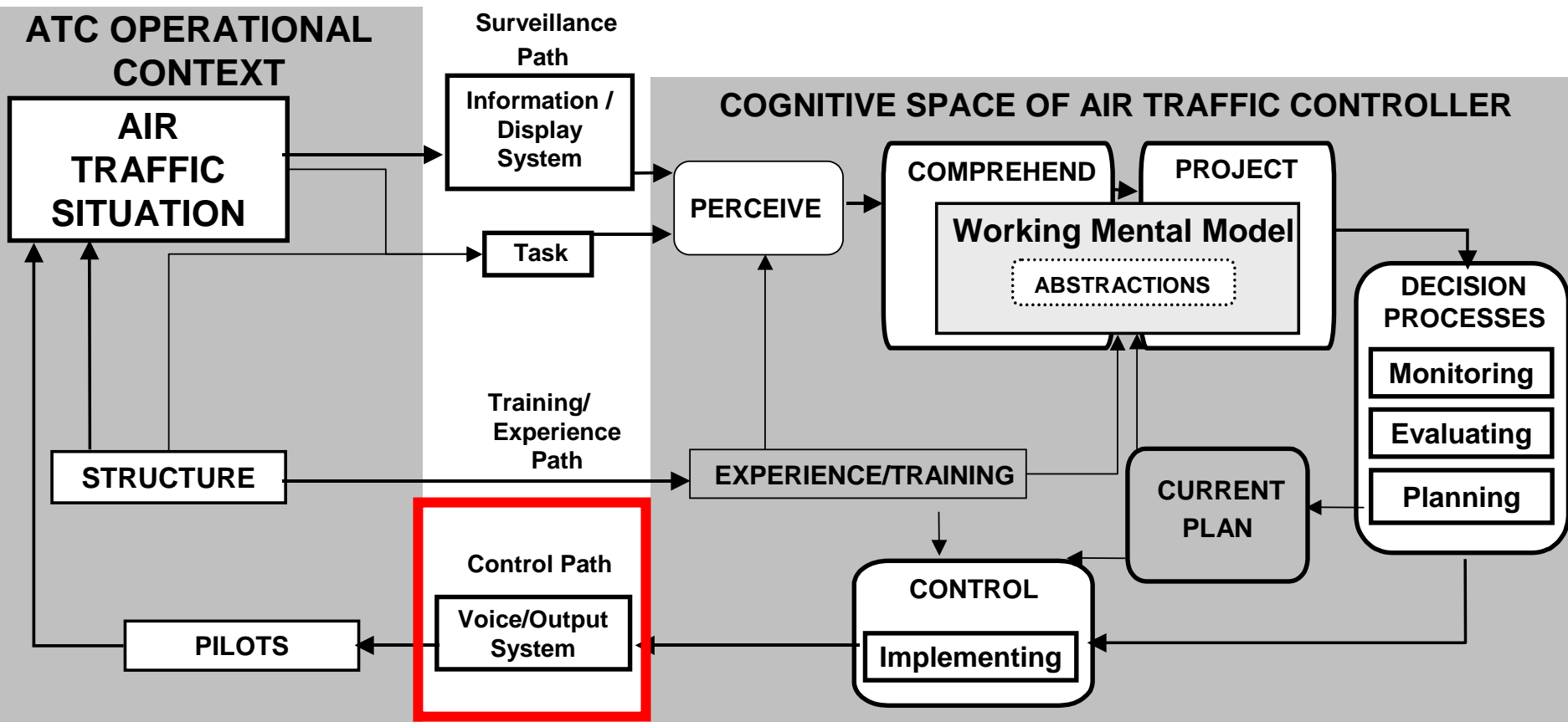


# Information/Display System

## Detailed View of Dynamic Information



# Control Path

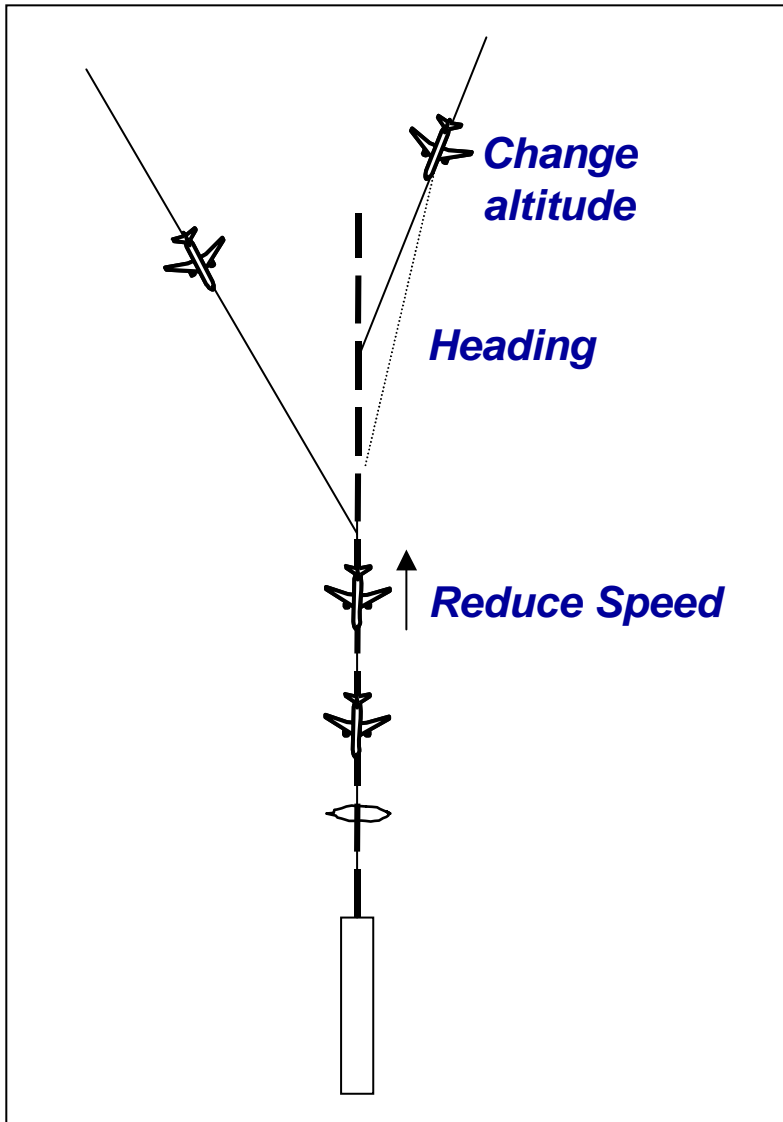




# Control Path

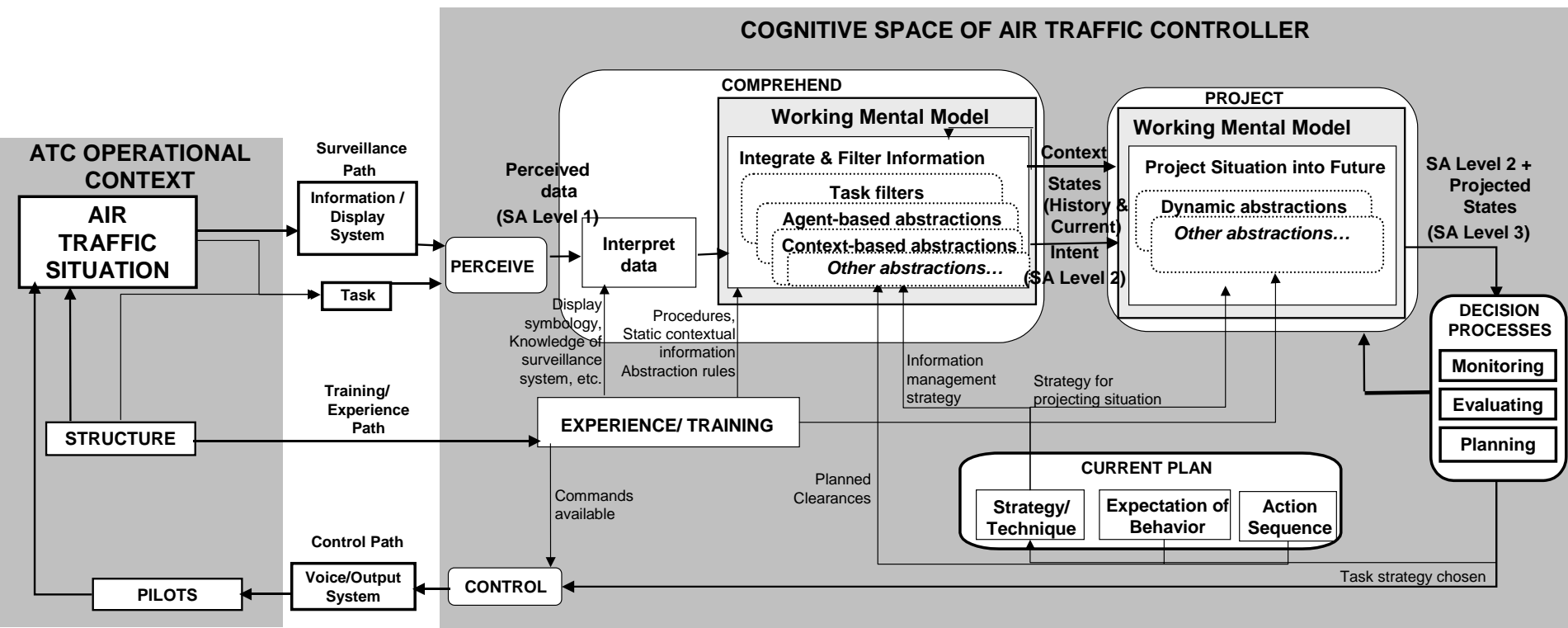
- ❑ Procedures → Control Command Availability → Cognitive Abstractions
- ❑ Limited set of commands allow controller to modify the evolution of the situation at different levels:
  - ❑ Position-based
    - Vertical (e.g., descend and maintain <altitude> feet)
    - Lateral (e.g., Turn left/right to <heading> degrees)
  - ❑ Velocity-based
    - Lateral (e.g., change speed to <kts>)
    - Vertical (e.g., expedite descent)
  - ❑ Trajectory-based (e.g., cleared ILS 4R)
  - ❑ Constraint
    - ❑ Temporal (e.g., ...until/after/before <time>)
    - ❑ Lateral (e.g., ...until <fix>)
    - ❑ Vertical (e.g., ...at/below/above <alt> ft)
    - ❑ Coordination (e.g., ...until advised by <unit>)
- ❑ System cycle time limits response to system (~30 sec for TRACON)
  - ❑ Pilot response time
  - ❑ Aircraft response time
  - ❑ Surveillance update
- ❑ Reduces intent uncertainty

# Final Approach Controllability

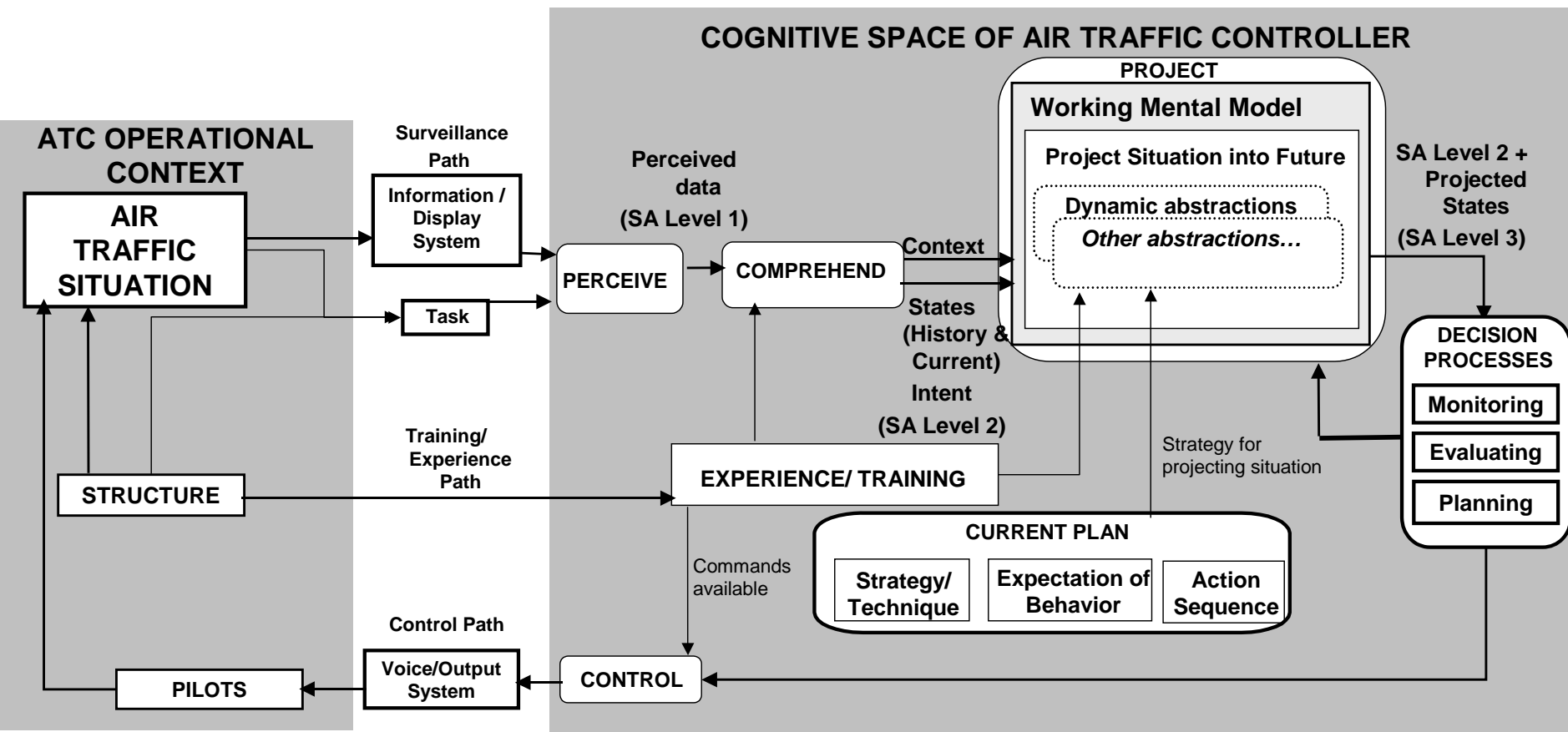


- Position-based (heading/ altitude) and velocity-based controls are used most frequently

# Comprehension & Projection



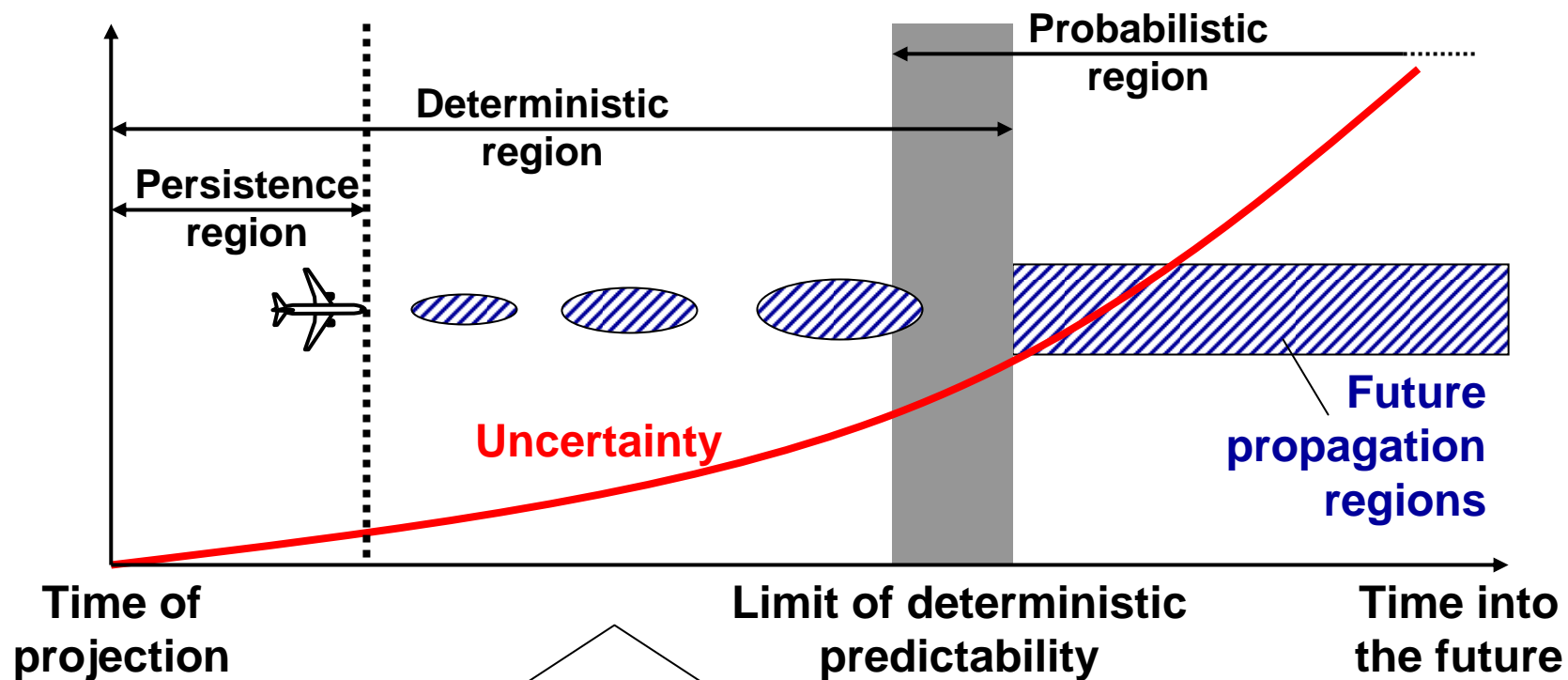
# Projection



- ❑ **Projection** is defined as the evolution of the mental model of the system into the future over the time required to execute and surveill a response to a command to keep the future behavior of the system within the task requirements



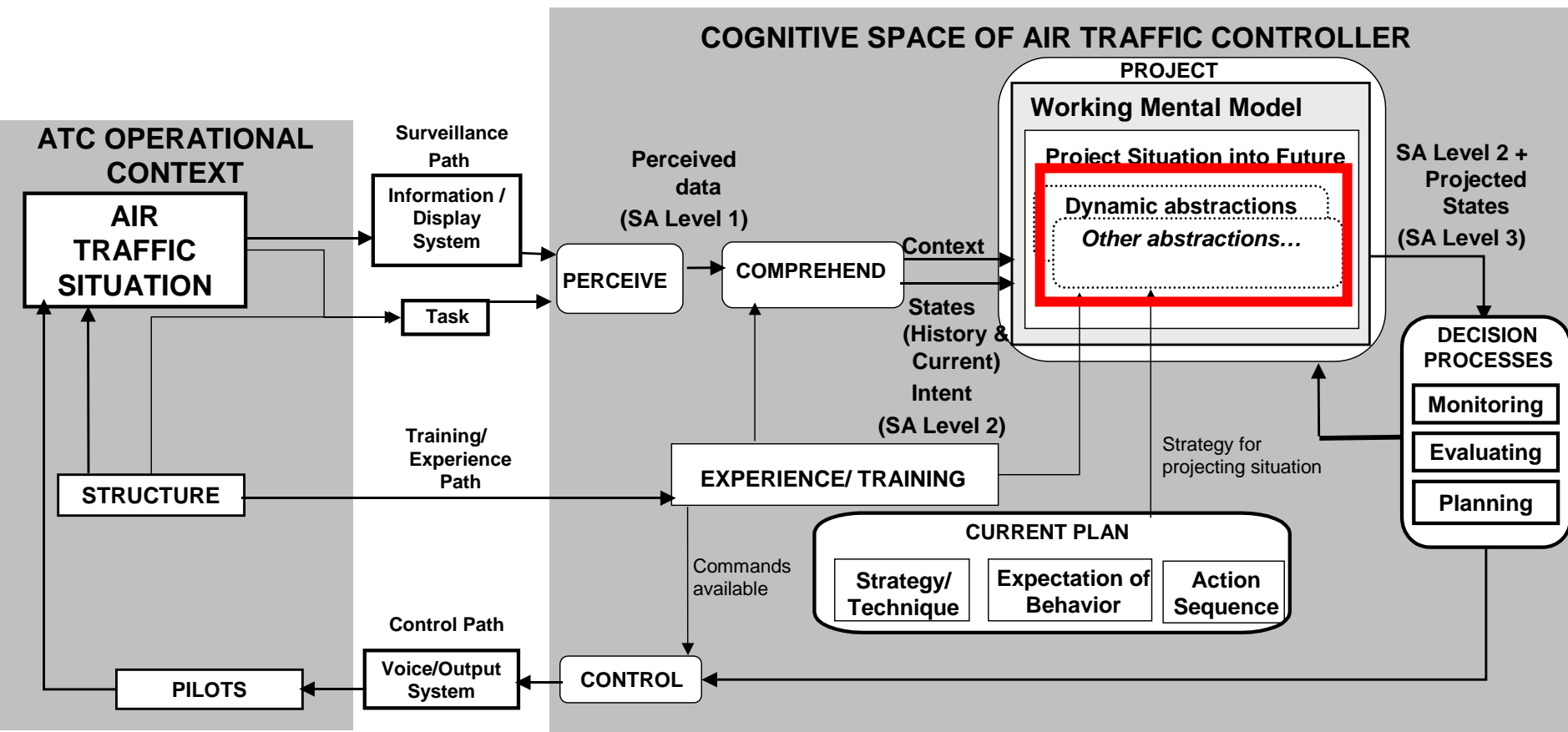
# Time of Projection



Task-based Projection Requirement

- Procedure
- Controllability

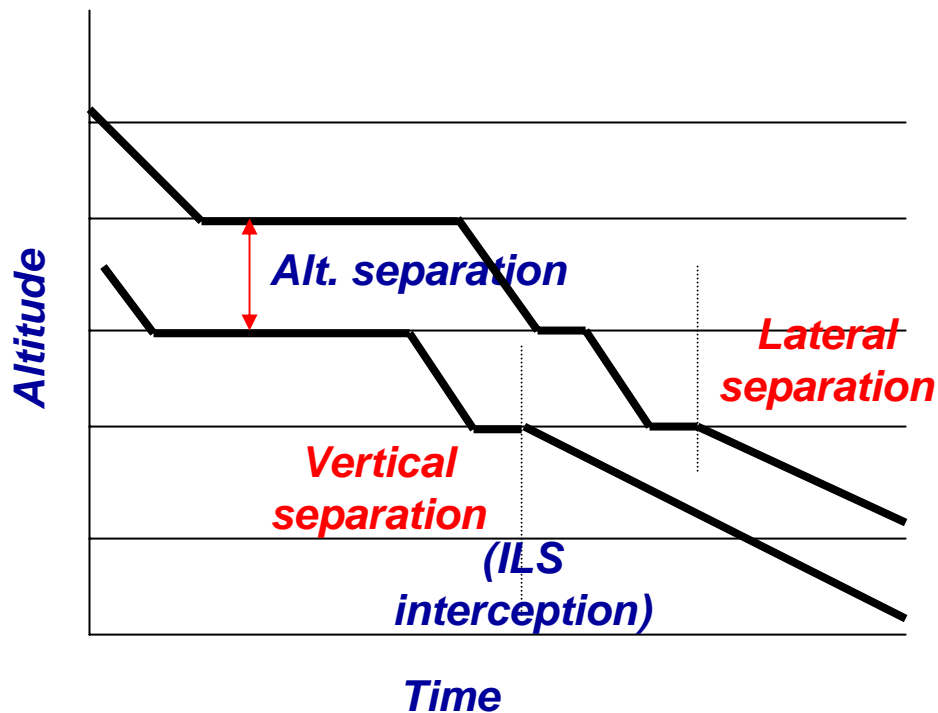
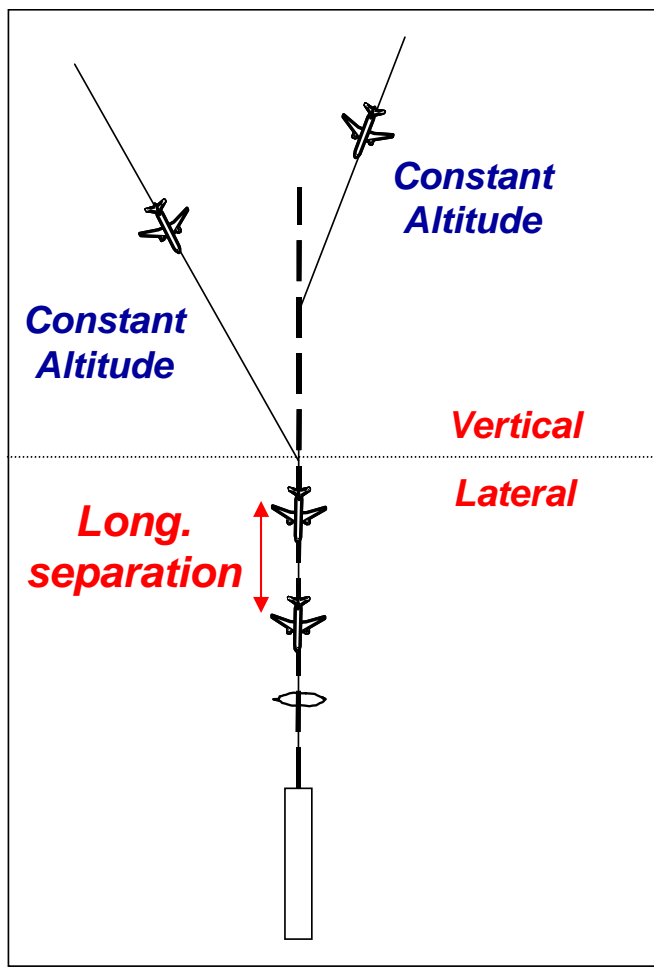
# Dynamic Abstractions



- ❑ Dynamic abstractions are the abstractions which support projection of the system dynamics, e.g.:
  - ❑ Constant Velocity
  - ❑ Constant Altitude

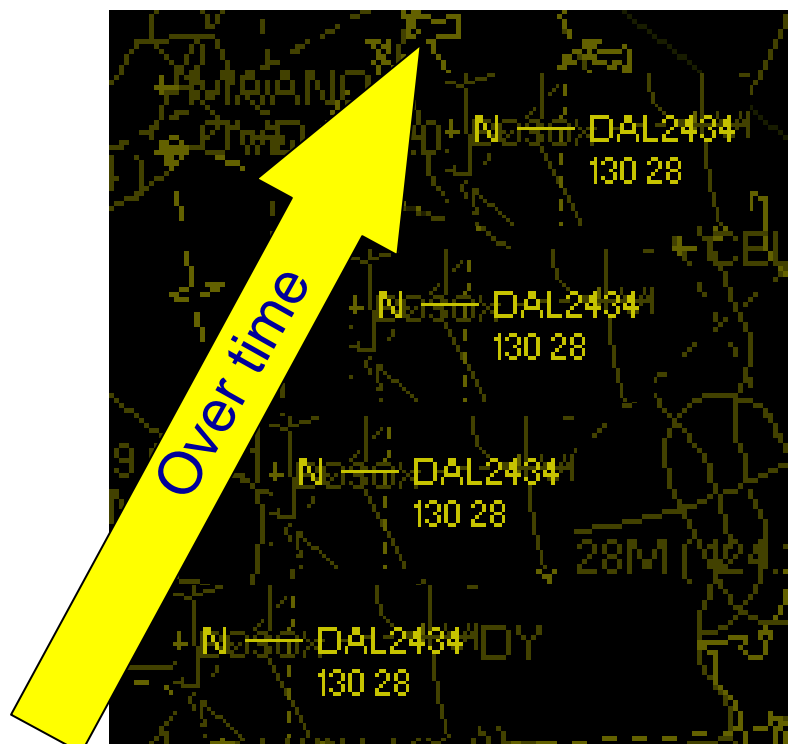
# Constant Altitude Abstraction

- Constant altitudes (CA) (achieved either through clearances or through procedures) ensures that merging traffic flows will be separated in at least the vertical dimension

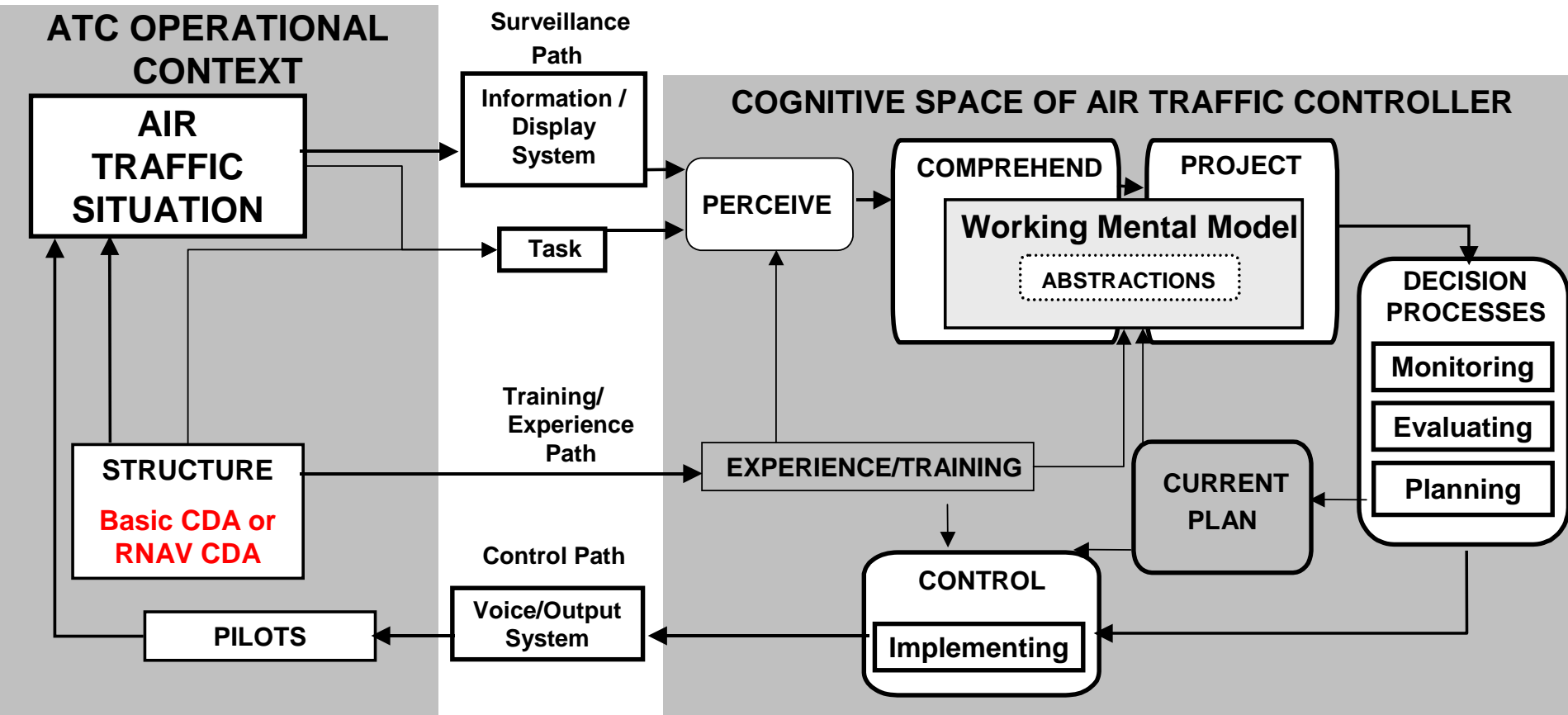


# Constant Velocity Abstraction

- ❑ Constant velocity (CV) is used as a way to establish a pattern to aid projection by equalizing distance traveled between updates
- ❑ If minimum lateral separation between 2 aircraft is reached, controllers can ensure this separation throughout the approach by commanding the aircraft to proceed at the same speed



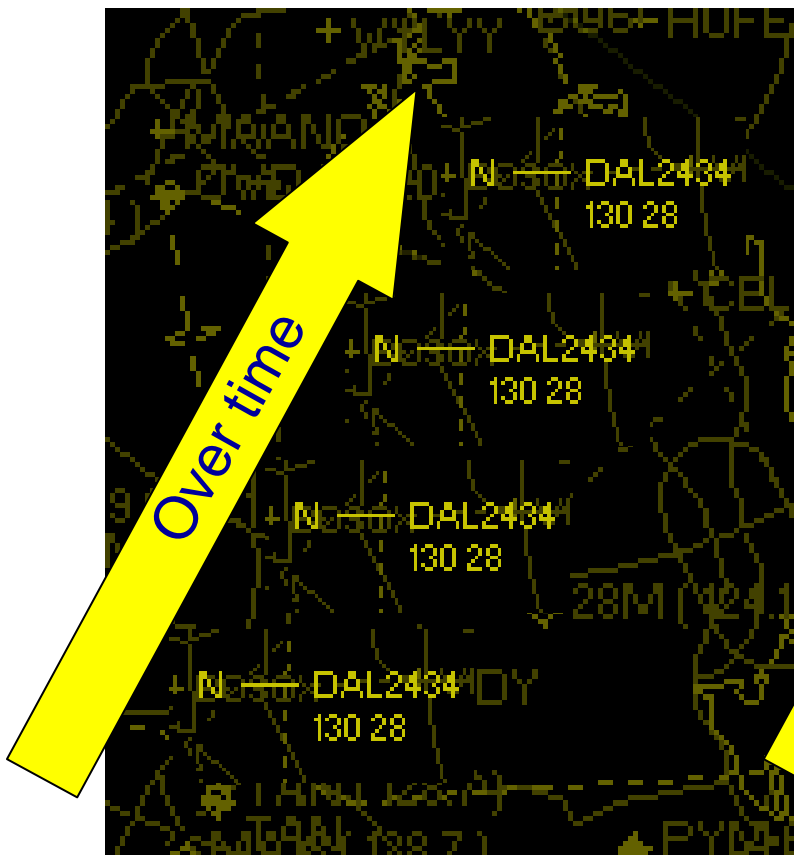
# Cognitive Differences between Conventional & CDA Procedures



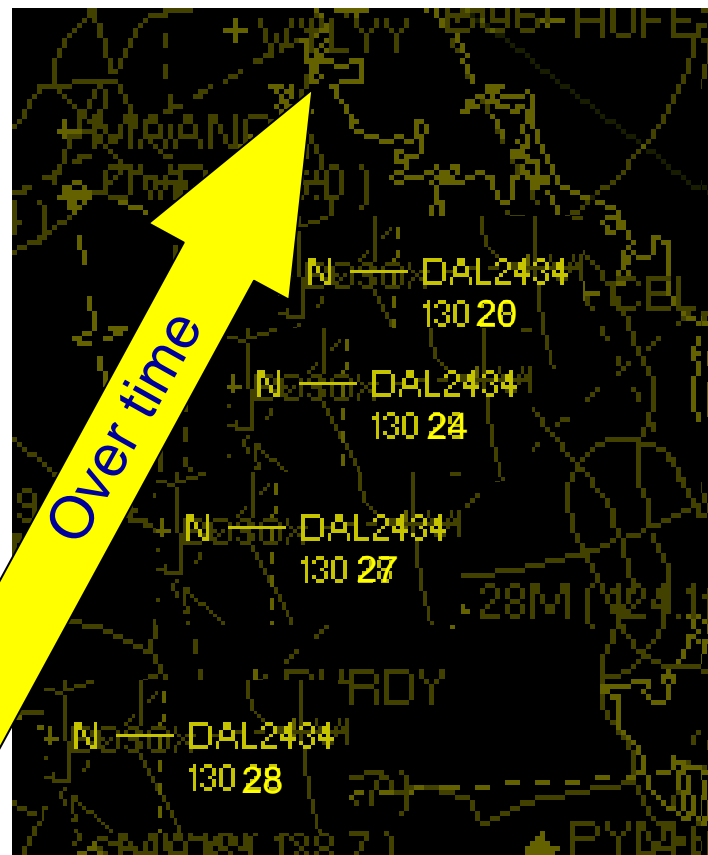
## □ Key Cognitive Differences

- Loss of abstractions (constant velocity & constant altitude)
- Reduction of controllability

# Constant Velocity Abstraction Lost

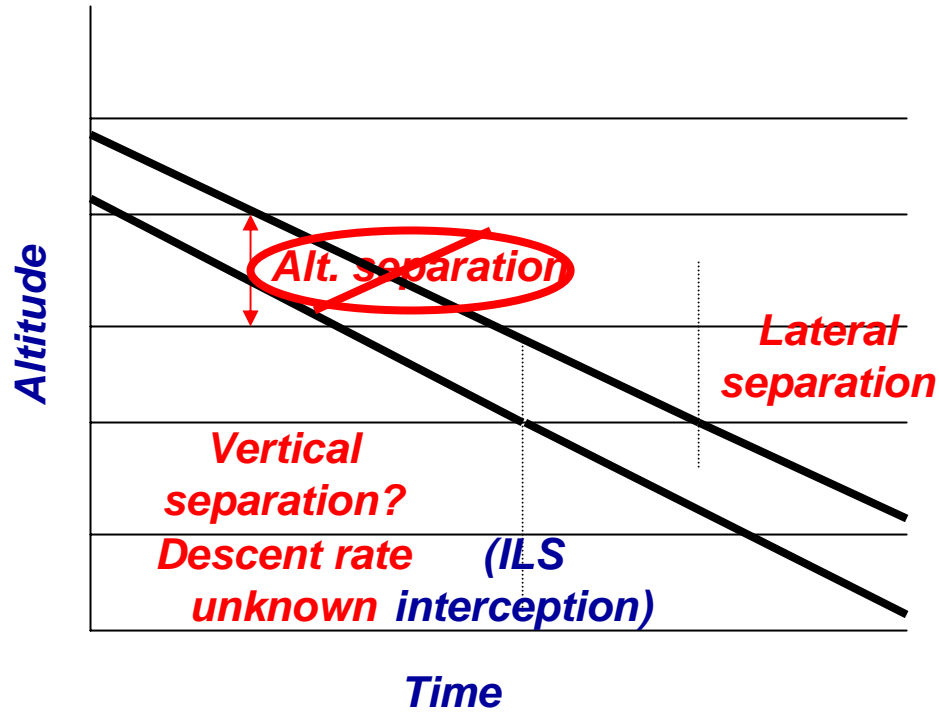
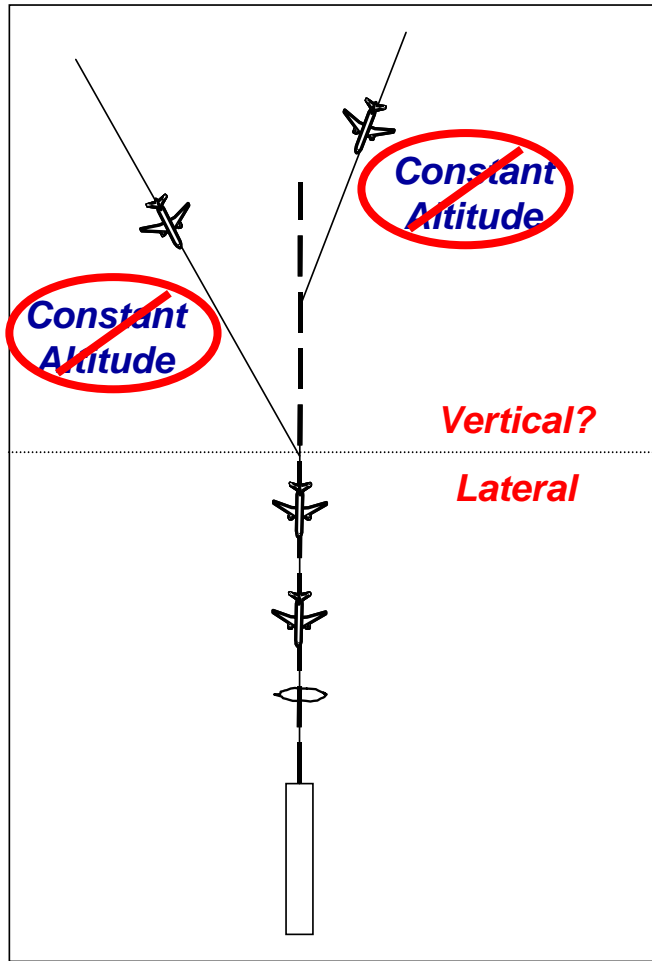


Constant velocity case

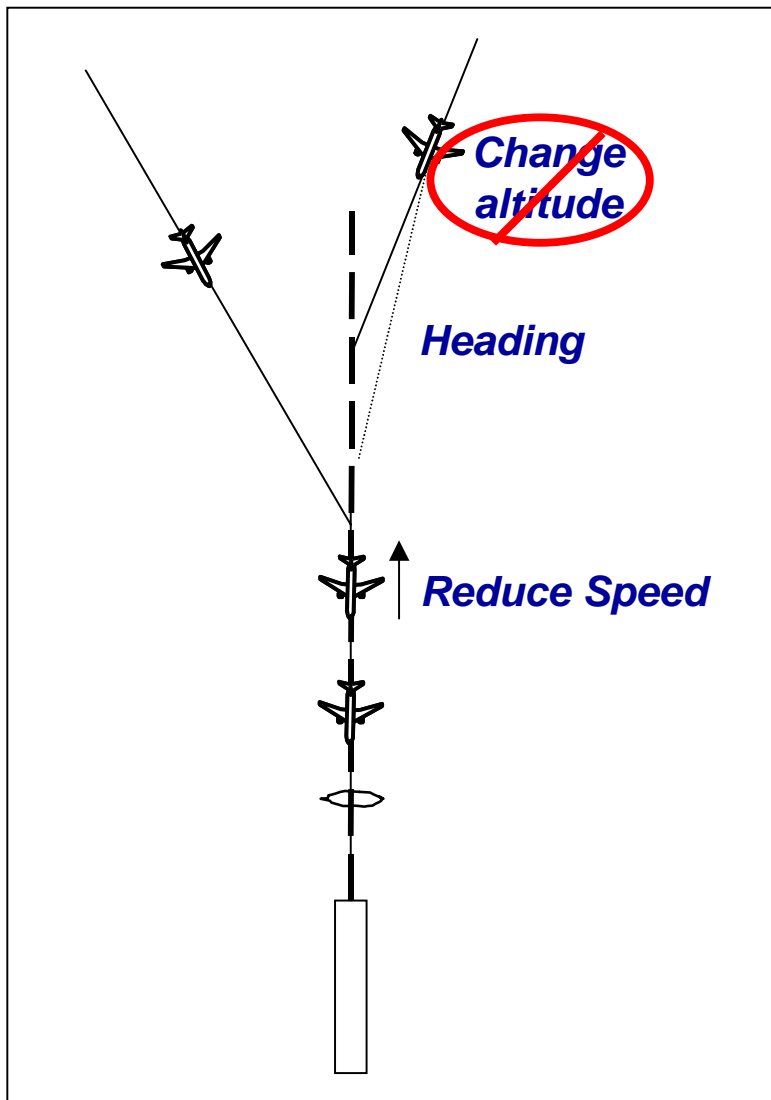


Decelerating case

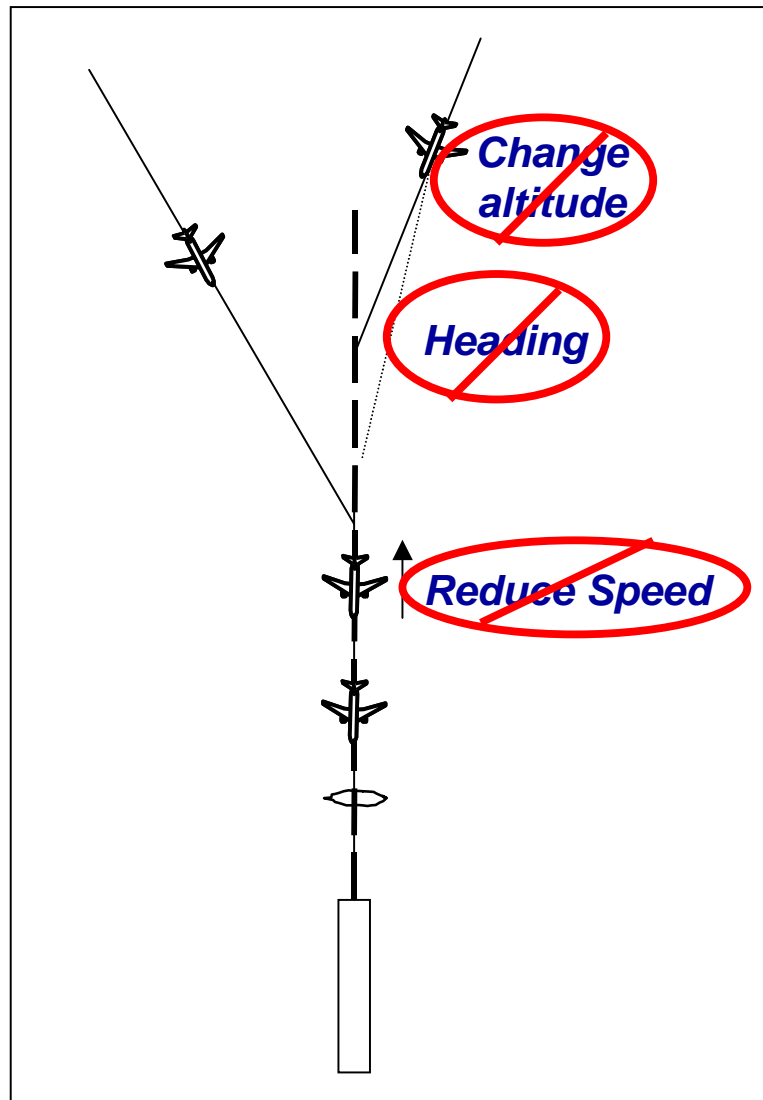
# Constant Altitude Abstraction Lost



# Controllability Differences



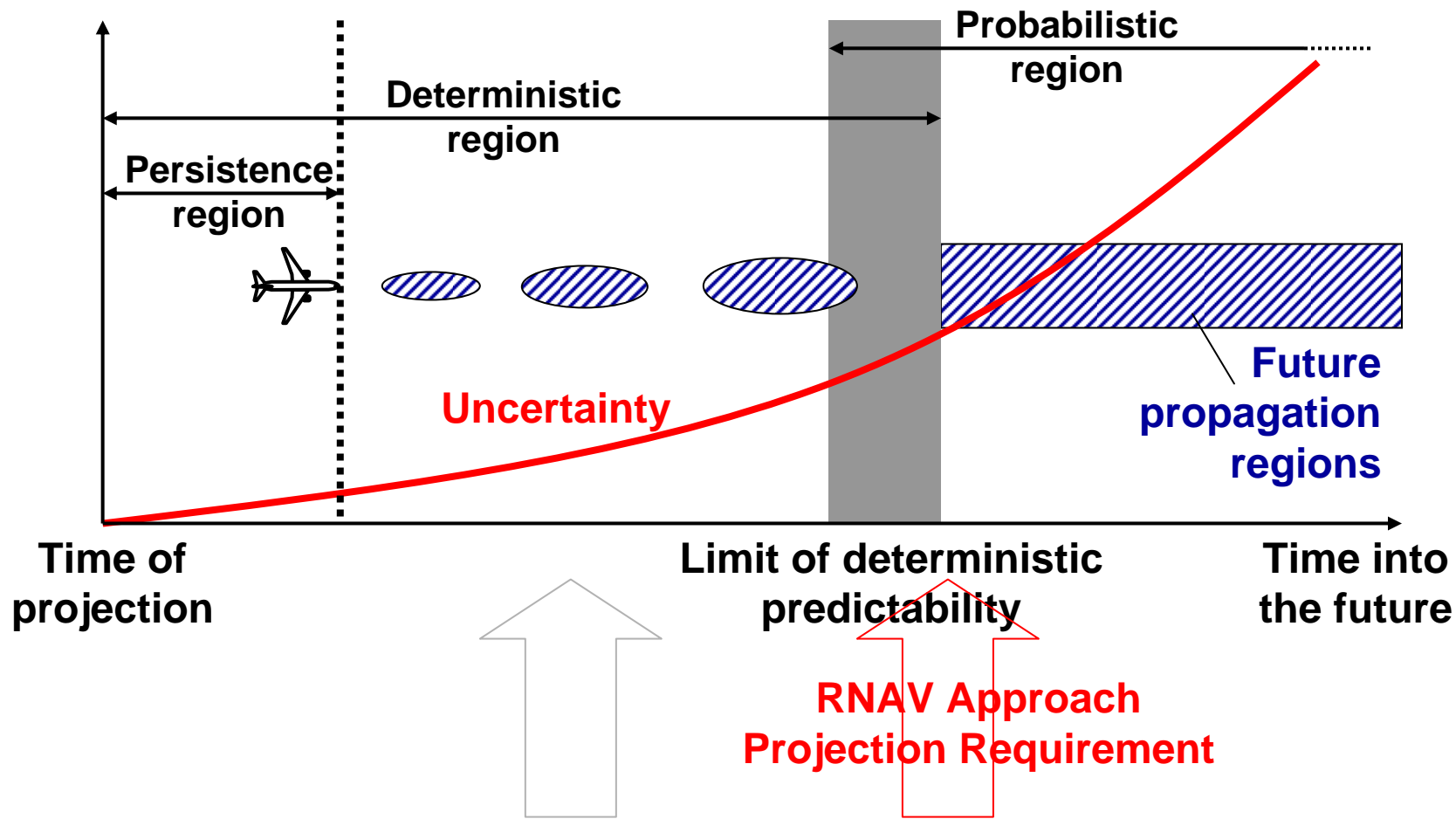
*Basic CDA*



*RNAV CDA*



# Time of Projection Changes in RNAV procedure



- Reducing controllability increases the timescale over which projection required, making projection more difficult



# Dynamic Differences

- Controller may substitute lost abstractions with more complicated abstractions
  - Aircraft are descending at different rates (Basic & RNAV CDAs)
  - Aircraft may be in speed transition over longer periods (RNAV CDA)
- Variability of dynamics in CDAs may also increase
  - Dynamics vary with track distance & aircraft type in Basic CDA and vary with aircraft type & FMS logic in RNAV CDA



# Workload Impacts in CDA Procedures

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## □ Basic CDA

- Track distance task is added
- Vertical projection task more complicated

## □ RNAV CDA

- Projection time into the future increases
- Tactical control decreases

□ Primary cognitive differences:

	<b>Basic CDA</b>	<b>RNAV CDA</b>
<b>Structure-based Abstractions</b>	Loss of Constant Altitude abstraction	Loss of Constant Altitude & Constant Velocity abstractions
<b>Controllability</b>	Loss of altitude controllability; Addition of Track Distance control	Loss of state (heading & altitude) and velocity controllability; Only “clear”/”abort” procedure
<b>Time into Future Req.</b>	No difference	Extended time into future projection requirement
<b>Complexity of dynamics</b>	Vertical complexity increases	Vertical & Longitudinal complexity increases; Lateral complexity decreases
<b>Variability of dynamics</b>	Vertical variability increases	Vertical & longitudinal variability increases; Lateral variability decreases
<b>Controller Workload</b>	May increase due to track distance estimations and vertical projection requirements	May increase due to requirement to project further into future due to lack of tactical controllability

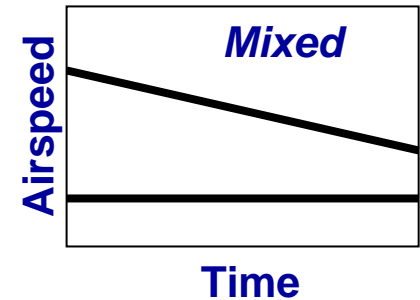
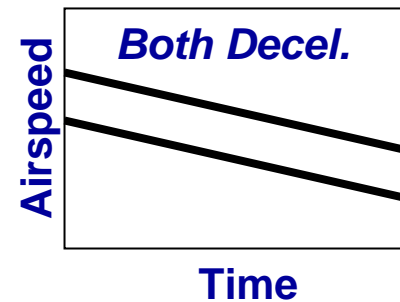
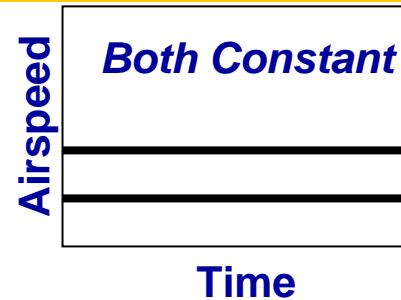


# Constant Velocity Structure Experiment

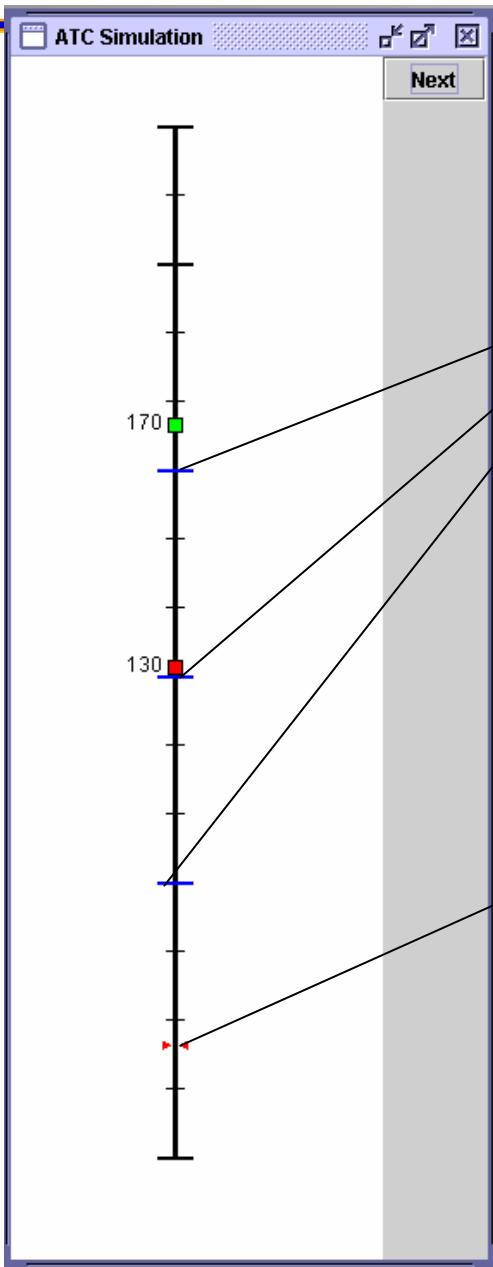
- ❑ Constant Velocity was identified as a key abstraction in the Cognitive Difference Analysis
- ❑ Can controllers create new abstractions to replace lost constant velocity abstraction?
- ❑ **Hypothesis:** Periods of constant speed are a key structure-based abstraction used in improving projection performance.
- ❑ **Goal:** Determine if some benefits provided by constant speed structure lost during low noise approach can be recovered by using standard deceleration profiles
- ❑ **Controllers' Task:** project the final separation of a pair of aircraft at different times, but **do not issue control commands**

# Independent Variables

- Deceleration profile:
  - Both constant speed
  - Mixed: One constant speed, one decelerating
  - Both decelerating
- Endspeed of aircraft
  - Aircraft 1 faster (opening case)
  - Aircraft 2 faster (closing case)
  - Same
- Final separation is counterbalanced across cases



# Task



- ❑ 3 projections of final separation must be made, each made by the time that Aircraft 1 passes a blue hash mark on the flight path
- ❑ Projection is recorded using red arrowheads



# Dependent Variables

- Accuracy of projection
  - Difference between projected separation & actual separation when aircraft 1 crosses the threshold
- Subjective rating of difficulty of constant versus decelerating aircraft projection and the strategy used to project separation





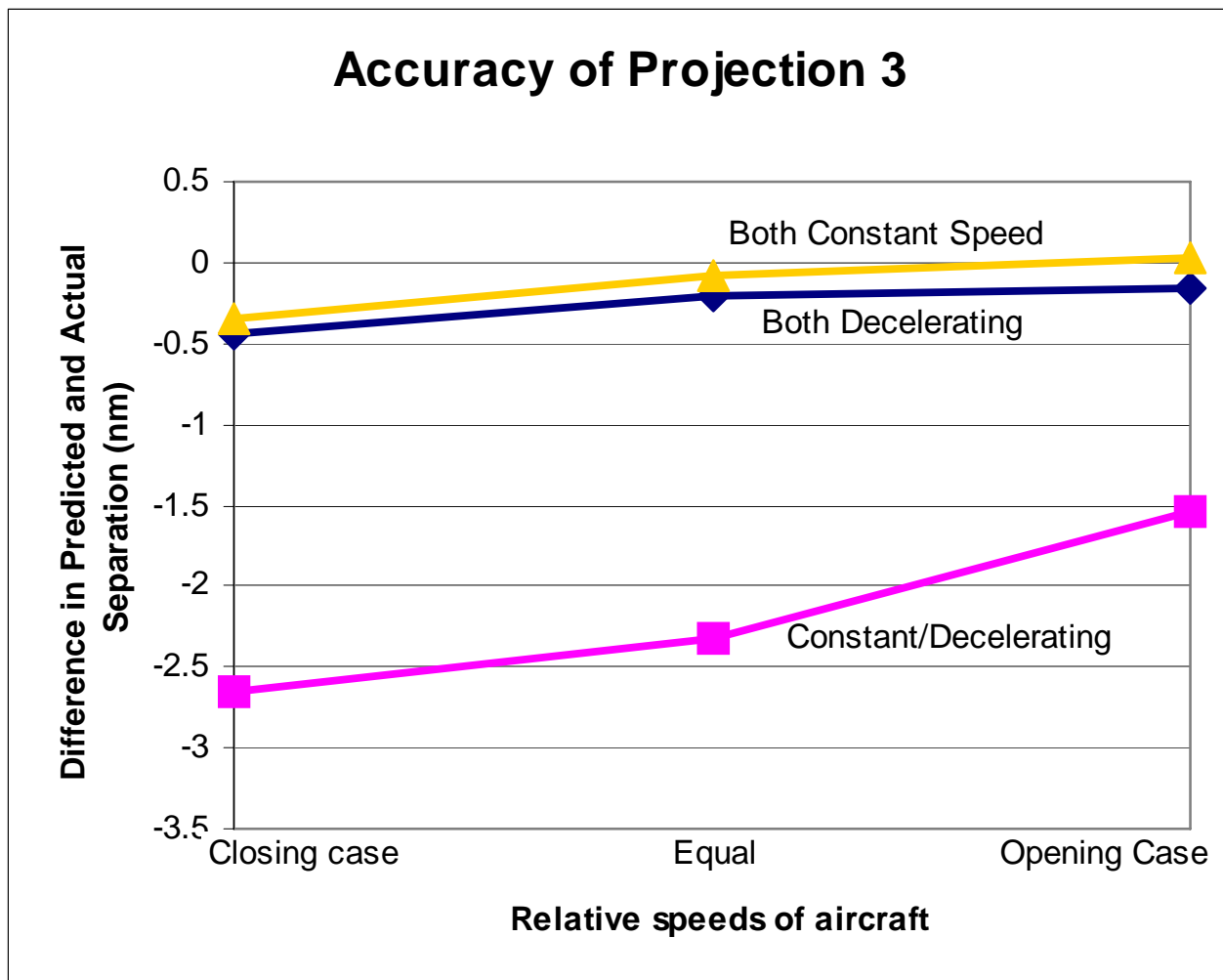
# Participants

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- ❑ 8 French student controllers with an average of 1.25 years experience
- ❑ 5 were Approach/Tower controllers
- ❑ 2 were En Route Center Controllers

# Accuracy between Speed Profiles

- Controllers projected less accurately in the mixed speed profile scenarios (closing case:  $t=2.021$ ,  $p<.05$ , equal case:  $t= 1.279$ ,  $p<.15$ )
- When both aircraft decelerated at the same rate, projection accuracy equaled the accuracy when both aircraft proceeded at constant speed





# Subjective Responses

- Difficulty of constant speed vs. deceleration
  - 6 of 8 said that decelerating was more difficult
  - One mentioned that the mixed profile opening case was the most difficult
- Strategy during the task:
  - *Heuristic:* 6 of 8 mentioned sampling the separation at two points then estimating separation based on the difference between the two samples
  - 2 mentioned missing the speed vector on the radar display



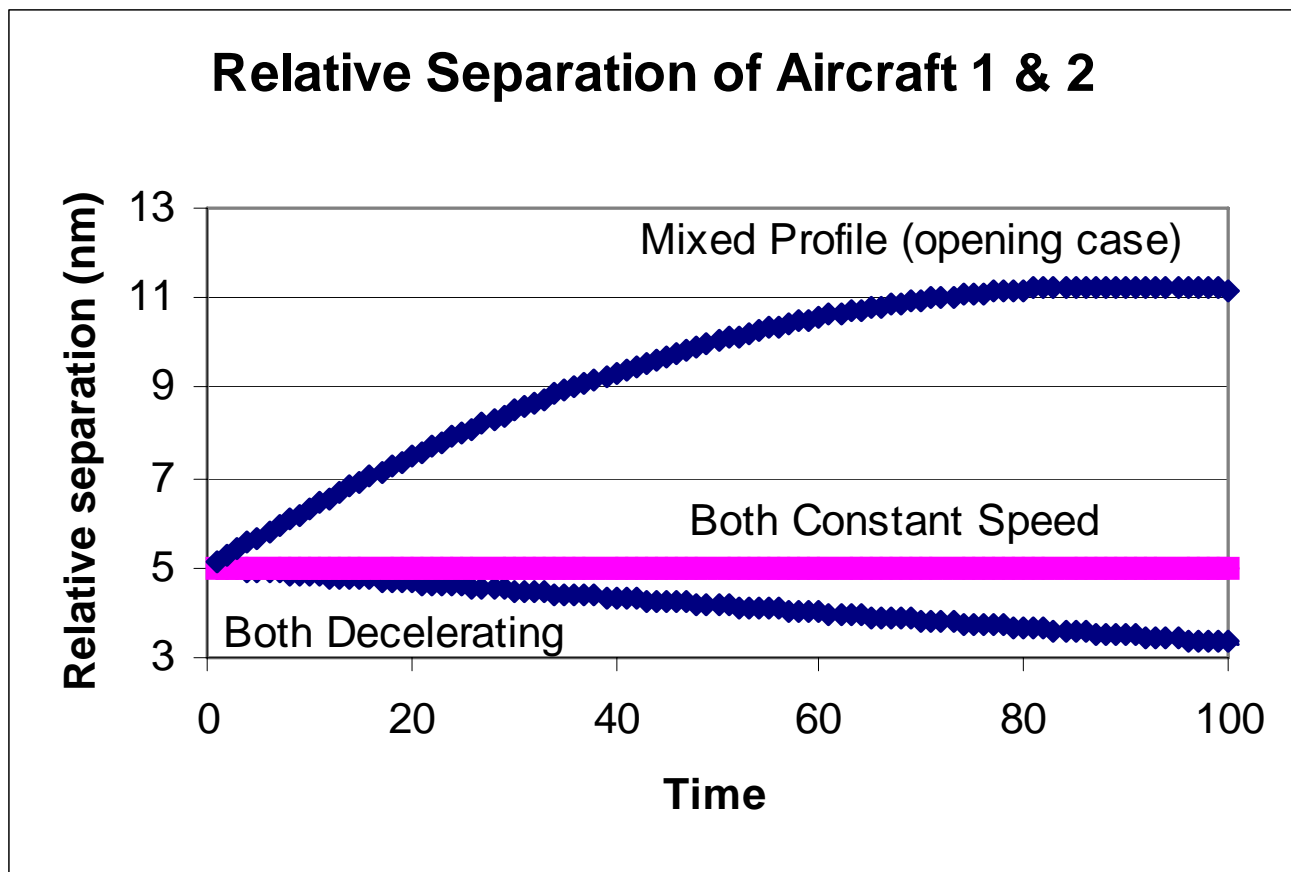
# Results Discussion

- ❑ Accuracy: Controllers were more accurate in projecting both constant or both decelerating aircraft than in projecting mixed profile aircraft
- ❑ A simple mental calculation based on separation sampling could be established for the constant & both decelerating case because the relative separation change over time was either constant or appeared linear
- ❑ Mixed profile scenarios: Possibility that no simple mental calculation could be established because the relative separation change was nonlinear



# Projecting Relative Separation

- The controllers' task in this experiment was to project relative separation between the two aircraft
- Relative separation in the Mixed Profile case was an observable nonlinear function, making the projection task more difficult





# Conclusions

- ❑ Controllers' acceptance & ability to project future behavior of aircraft on approach are a barrier to implementing low noise procedures
- ❑ Key differences between procedures affect cognitive processes:
  - Loss of simple dynamic abstractions → More complex dynamics to project & higher workload
  - Loss of controllability → Longer projection time required
  - Impact on workload due to changed tasks & projection requirements
- ❑ ATC support is required, possibly in the form of:
  - Reduction of projection requirement
    - E.g., Improving ATC speed controllability in RNAV CDA procedure-speed commands and/or speedbrakes control
  - Supporting the formation of new projection abstractions
    - E.g., Increasing predictability of dynamics- structured deceleration profiles